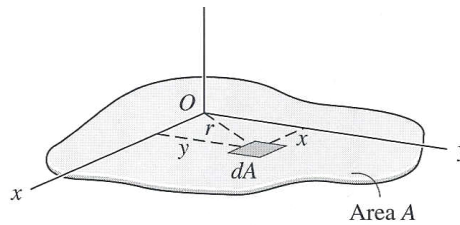


Moment of Inertia and Properties of Plane Areas

The **Moment of Inertia (I)** is a term used to describe the capacity of a cross-section to resist bending. It is always considered with respect to a reference axis such as X-X or Y-Y. It is a mathematical property of a section concerned with a surface area and how that area is distributed about the reference axis (*axis of interest*). The reference axis is usually a centroidal axis.

The moment of inertia is also known as the **Second Moment of the Area** and is expressed mathematically as:



$$I_x = \int_A y^2 dA$$

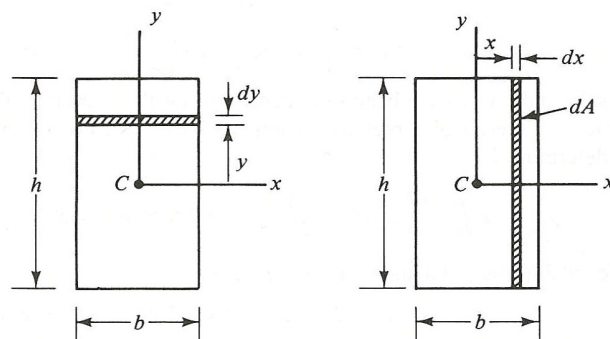
$$I_y = \int_A x^2 dA$$

Where

y = distance from the x axis to area dA

x = distance from the y axis to area dA

Example



Radius of Gyration:

$$k_x = \sqrt{\frac{I_x}{A}}$$

$$k_y = \sqrt{\frac{I_y}{A}}$$

The **radius of gyration** of an area with respect to a particular axis is the square root of the quotient of the moment of inertia divided by the area. It is the distance at which the entire area must be assumed to be concentrated in order that the product of the area and the square of this distance will equal the moment of inertia of the actual area about the given axis. In other words, the radius of gyration describes the way in which the total cross-sectional area is distributed around its centroidal axis. If more area is distributed further from the axis, it will have greater resistance to buckling. The most efficient column section to resist buckling is a circular pipe, because it has its area distributed as far away as possible from the centroid.

Rearranging we have:

$$I_x = k_x^2 A$$

$$I_y = k_y^2 A$$

The radius of gyration is the distance k away from the axis that all the area can be concentrated to result in the same moment of inertia.

Polar Moment of Inertia:

$$I_p = \int_A \rho^2 dA$$

$$I_p = \int_A (x^2 + y^2) dA$$

$$I_p = \int_A x^2 dA + \int_A y^2 dA$$

$$I_p = I_x + I_y$$

In many texts, the symbol J will be used to denote the polar moment of inertia.

$$J = I_x + I_y$$

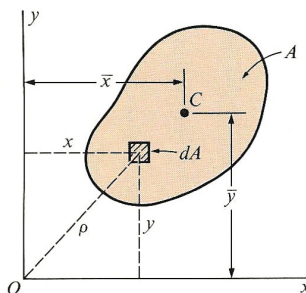
Shear stress formula

$$\tau = \frac{Tr}{J}$$

Product of Inertia:

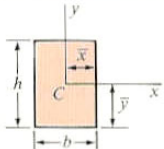
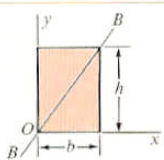
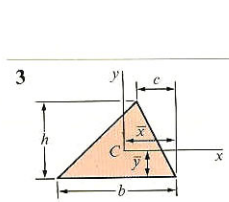
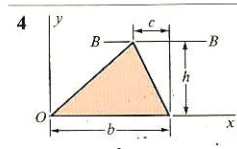
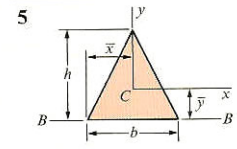
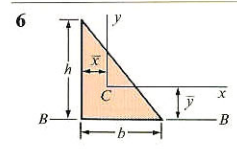
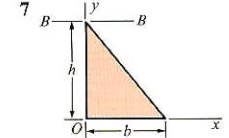
$$I_{xy} = \int_A xy dA$$

Consider the following:

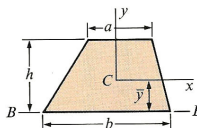
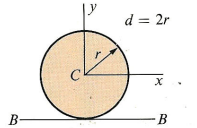
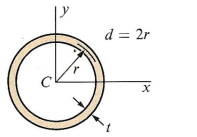
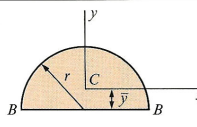
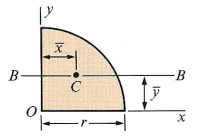
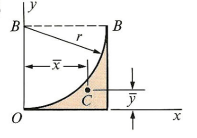


If an area has at least one axis of symmetry, the product of inertia is zero.

Properties of Plane Areas

<p>1</p> 	<p>Rectangle (Origin of axes at centroid.)</p> $A = bh \quad \bar{x} = \frac{b}{2} \quad \bar{y} = \frac{h}{2}$ $I_x = \frac{bh^3}{12} \quad I_y = \frac{hb^3}{12}$ $I_{xy} = 0 \quad I_p = \frac{bh}{12}(h^2 + b^2)$
<p>2</p> 	<p>Rectangle (Origin of axes at corner.)</p> $I_x = \frac{bh^3}{3} \quad I_y = \frac{hb^3}{3}$ $I_{xy} = \frac{b^2h^2}{4} \quad I_p = \frac{bh}{3}(h^2 + b^2) \quad I_{BB} = \frac{b^3h^3}{6(b^2 + h^2)}$
<p>3</p> 	<p>Triangle (Origin of axes at centroid.)</p> $A = \frac{bh}{2} \quad \bar{x} = \frac{b+c}{3} \quad \bar{y} = \frac{h}{3}$ $I_x = \frac{bh^3}{36} \quad I_y = \frac{bh}{36}(b^2 - bc + c^2)$ $I_{xy} = \frac{bh^2}{72}(b - 2c) \quad I_p = \frac{bh}{36}(h^2 + b^2 - bc + c^2)$
<p>4</p> 	<p>Triangle (Origin of axes at vertex.)</p> $I_x = \frac{bh^3}{12} \quad I_y = \frac{bh}{12}(3b^2 - 3bc + c^2)$ $I_{xy} = \frac{bh^2}{24}(3b - 2c) \quad I_{BB} = \frac{bh^3}{4}$
<p>5</p> 	<p>Isosceles triangle (Origin of axes at centroid.)</p> $A = \frac{bh}{2} \quad \bar{x} = \frac{b}{2} \quad \bar{y} = \frac{h}{3}$ $I_x = \frac{bh^3}{36} \quad I_y = \frac{hb^3}{48} \quad I_{xy} = 0$ $I_p = \frac{bh}{144}(4h^2 + 3b^2) \quad I_{BB} = \frac{bh^3}{12}$ <p>(Note: For an equilateral triangle, $h = \sqrt{3}b/2$.)</p>
<p>6</p> 	<p>Right triangle (Origin of axes at centroid.)</p> $A = \frac{bh}{2} \quad \bar{x} = \frac{b}{3} \quad \bar{y} = \frac{h}{3}$ $I_x = \frac{bh^3}{36} \quad I_y = \frac{hb^3}{36} \quad I_{xy} = -\frac{b^2h^2}{72}$ $I_p = \frac{bh}{36}(h^2 + b^2) \quad I_{BB} = \frac{bh^3}{12}$
<p>7</p> 	<p>Right triangle (Origin of axes at vertex.)</p> $I_x = \frac{bh^3}{12} \quad I_y = \frac{hb^3}{12} \quad I_{xy} = \frac{b^2h^2}{24}$ $I_p = \frac{bh}{12}(h^2 + b^2) \quad I_{BB} = \frac{bh^3}{4}$

Properties of Plane Areas

<p>8</p> 	<p>Trapezoid (Origin of axes at centroid.)</p> $A = \frac{h(a+b)}{2} \quad \bar{y} = \frac{h(2a+b)}{3(a+b)}$ $I_x = \frac{h^3(a^2 + 4ab + b^2)}{36(a+b)} \quad I_{BB} = \frac{h^3(3a+b)}{12}$
<p>9</p> 	<p>Circle (Origin of axes at center.)</p> $A = \pi r^2 = \frac{\pi d^2}{4} \quad I_x = I_y = \frac{\pi r^4}{4} = \frac{\pi d^4}{64}$ $I_{xy} = 0 \quad I_p = \frac{\pi r^4}{2} = \frac{\pi d^4}{32} \quad I_{BB} = \frac{5\pi r^4}{4} = \frac{5\pi d^4}{64}$
<p>10</p> 	<p>Circular ring (Origin of axes at center.) Approximate formulas for case when t is small.</p> $A = 2\pi r t = \pi d t \quad I_x = I_y = \pi r^3 t = \frac{\pi d^3 t}{8}$ $I_{xy} = 0 \quad I_p = 2\pi r^3 t = \frac{\pi d^3 t}{4}$
<p>11</p> 	<p>Semicircle (Origin of axes at centroid.)</p> $A = \frac{\pi r^2}{2} \quad \bar{y} = \frac{4r}{3\pi}$ $I_x = \frac{(9\pi^2 - 64)r^4}{72\pi} \approx 0.1098r^4 \quad I_y = \frac{\pi r^4}{8}$ $I_{xy} = 0 \quad I_{BB} = \frac{\pi r^4}{8}$
<p>12</p> 	<p>Quarter circle (Origin of axes at center of circle.)</p> $A = \frac{\pi r^2}{4} \quad \bar{x} = \bar{y} = \frac{4r}{3\pi}$ $I_x = I_y = \frac{\pi r^4}{16} \quad I_{xy} = \frac{r^4}{8}$ $I_{BB} = \frac{(9\pi^2 - 64)r^4}{144\pi} \approx 0.05488r^4$
<p>13</p> 	<p>Quarter-circular spandrel (Origin of axes at vertex.)</p> $A = \left(1 - \frac{\pi}{4}\right)r^2$ $\bar{x} = \frac{2r}{3(4-\pi)} \approx 0.7766r \quad \bar{y} = \frac{(10-3\pi)r}{3(4-\pi)} \approx 0.2234r$ $I_x = \left(1 - \frac{5\pi}{16}\right)r^4 \approx 0.01825r^4 \quad I_y = I_{BB} = \left(\frac{1}{3} - \frac{\pi}{16}\right)r^4 \approx 0.1370r^4$

Parallel Axis Theorem:

$$I_x = I_{xc} + Ad^2$$

$$I_y = I_{yc} + Ad^2$$

The moment of inertia of an area with respect to any given axis is equal to the moment of inertia with respect to the centroidal axis plus the product of the area and the square of the distance between the 2 axes.

The parallel axis theorem is used to determine the moment of inertia of composite sections.

Example

